

Crash Dynamics Summary



Federal Aviation
Administration

*Presented to: FAA/EASA/Industry
Composite Transport Fatigue, Damage
Tolerance, Maintenance and
Crashworthiness Workshop*

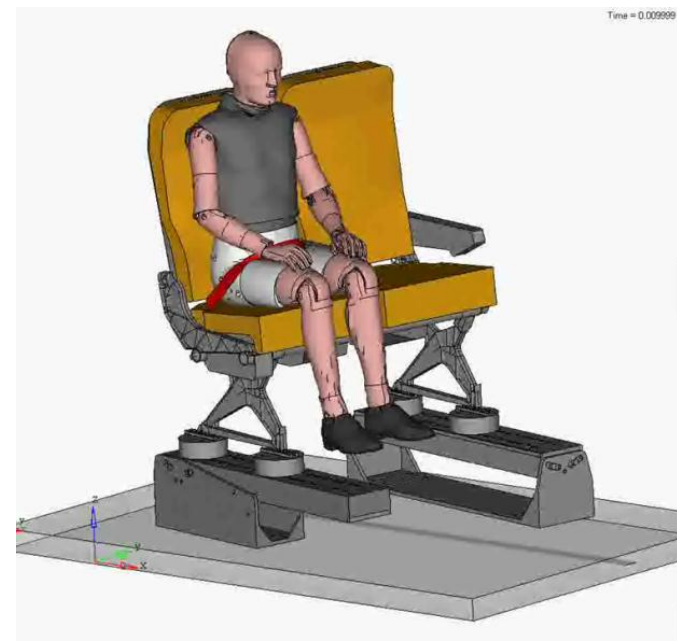
*By: Joseph Pelletiere, Chief Scientific
and Technical Advisor for Crash
Dynamics*

Date: 19 May 2011



Overview

- **History**
 - Provide basis of regulations
- **Modeling and Simulation**
 - How to apply M&S to regulations



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Modeling and Simulation

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Crashworthiness Requirements

- **No specific dynamic requirement for airplane level crashworthiness**
- **Demonstrate equivalent level of safety**
- **Impact conditions up to 30 ft/sec**
- **Passenger load**
 - 2/3
 - Maximum
- **Requirements on the seat performance**

Dynamic Impact Standards

- **Requirements on the seat performance**
- **Developed from**
 - Accident data
 - Parametric studies
 - Existing guidelines
 - FAA/NASA research
- **Provide occupant safety metrics**
- **Typically met through testing**
 - Modeling and simulation is an option

Dynamic Impact Standards

– Test 1

- **Combined Vertical/Longitudinal**

- Velocity change not less than 35 fps

- Vertical 30.3 fps
- Longitudinal 17.5 fps
- Peak Deceleration 14 G's minimum

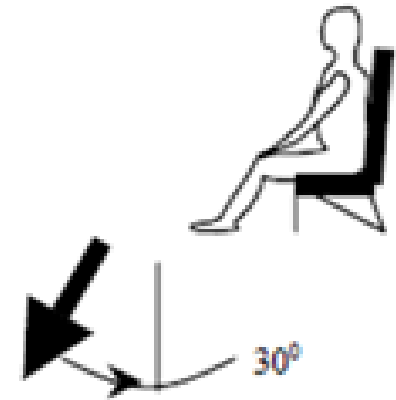
- Rise time = 0.08 sec

- Floor deformation

- None

- Evaluates spinal loads and injury

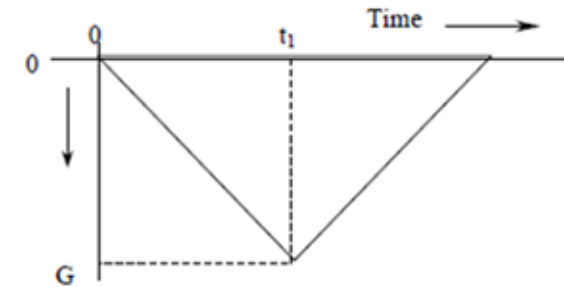
Test 1



Test Pulse simulating
Aircraft Floor
Deceleration -Time
History

Deceleration

t_1 = Rise time
 V_1 = Impact velocity



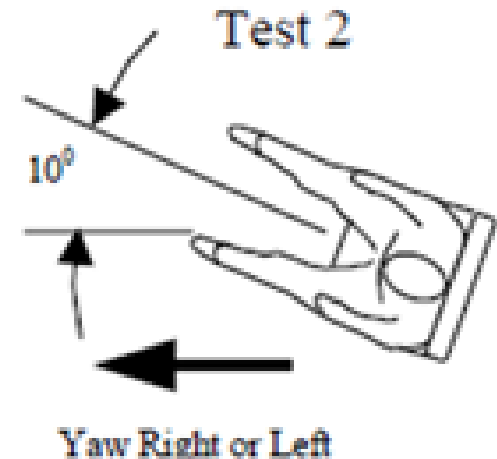
The ideal pulse is a symmetrical isosceles triangle

Dynamic Impact Standards

– Test 2

- **Longitudinal**

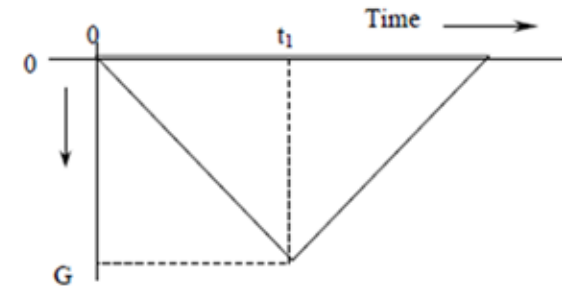
- Velocity change not less than 44 fps
 - Peak deceleration 16 G's minimum
- Rise time = 0.09 sec
- Floor deformation
 - 10° pitch
 - 10° roll
- Assess occupant restraint system
- Assess seat structural performance



Test Pulse simulating
Aircraft Floor
Deceleration -Time
History

Deceleration

t_1 = Rise time
 V_1 = Impact velocity



The ideal pulse is a symmetrical isosceles triangle

Factors in Crash Survivability

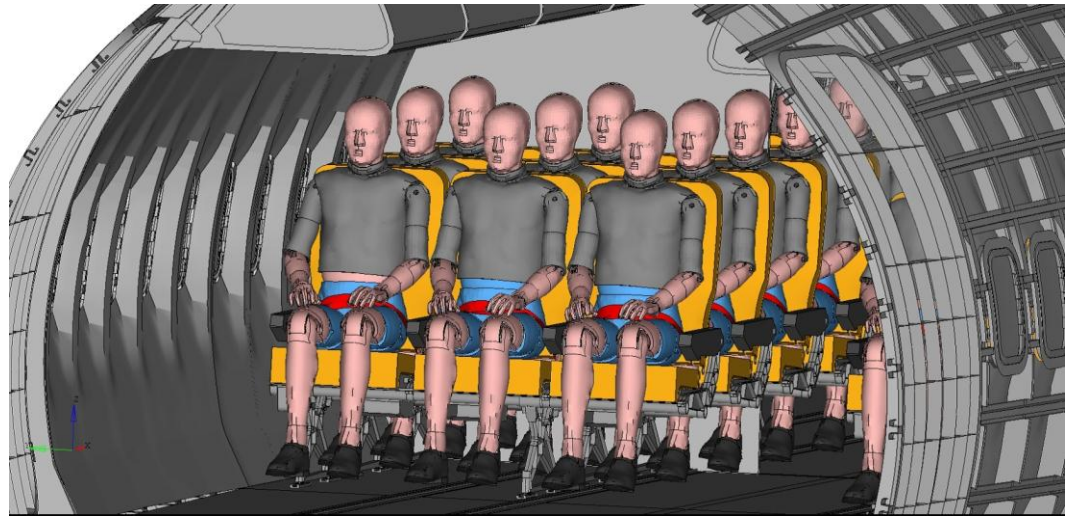
- **Retention of items of mass**
- **Maintenance of occupant emergency egress paths**
- **Maintenance of acceptable acceleration and loads experienced by the occupant**
- **Maintenance of survivable volume**

Occupant Focus

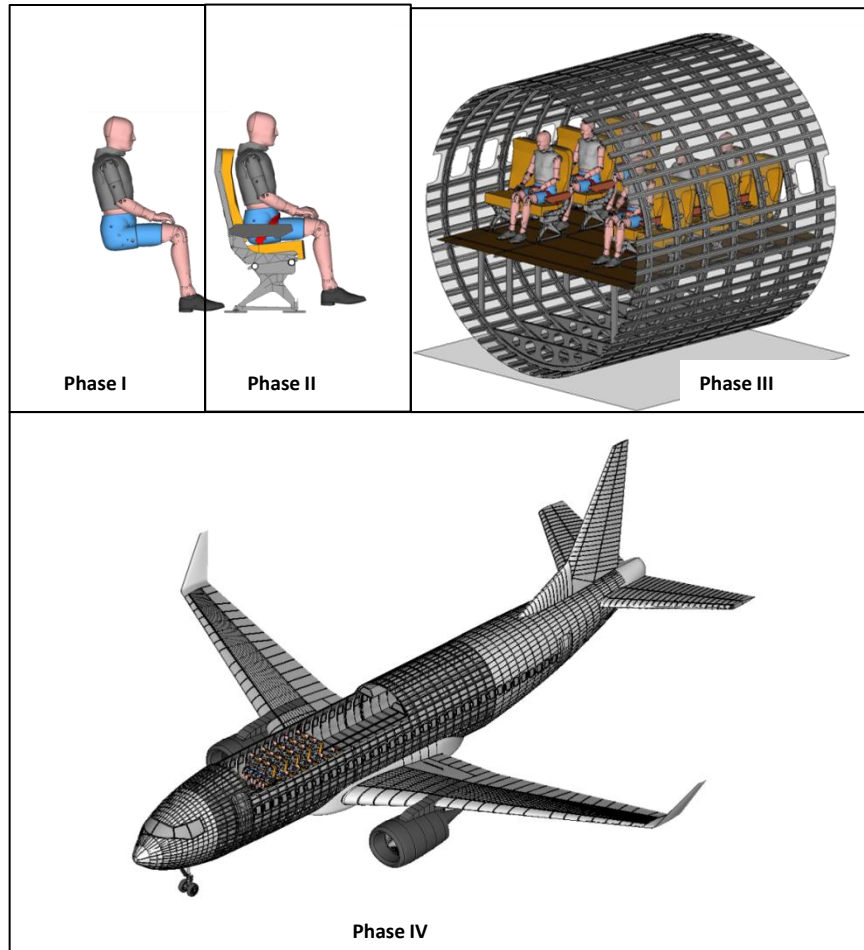
- **Four factors – prevention of injury to occupants**
- **Relate to space and energy management around the occupant**
- **Can be met by occupant/seat interface**
 - Assumes a minimum amount of energy management input

Modeling & Simulation

- **Can be used for new designs**
- **Will require demonstration of ELOS**
 - Demonstrate factors
- **Will require testing to support validation**
 - Drop Test
 - Components




Crashworthiness “Inside-Out Method”



- Phase 0: Define Occupant Injury Limits | FAR *.562 | ✓
- Phase I: Develop and validate occupant ATD numerical models |CBA I Part I : Experimental and Computational| SAE ARP 5765 | ✓
- Phase II: Define Modeling and Certification by Analysis Processes of Aerospace Seat Structures and Installations |AC 20-146|CBA I Part II: Experimental and Computational| SAE ARP 5765 ✓
- Phase III: Define Crashworthiness Requirements for Aircraft Structures |CBA II : Computational and Experimental/Accident Data Analysis|
- Phase IV: Define Structural CBA Methodology |CBA II : Computational and Experimental Procedures|

Modeling & Simulation

- **AC 20-146**
 - Methodology for Dynamic Seat Certification by Analysis
- **SAE ARP 5765** 
 - Analytical Methods for Aircraft Seat Design and Evaluation
- **Numerical Dummy Validation**
- **Best Practices Guide**

AC 20-146

- **Purpose: Acceptable means to demonstrate compliance with computer analysis validated by **DYNAMIC TESTS****
- **23/25/27/29.562**
- **Defines**
 - Minimum validation guidelines
 - Minimum documentation requirements
 - Conditions for use
 - Limitations

Modeling in Support of Dynamic Testing

- **Determination of worst-case for a seat design**
- **Determination of worst-case scenario for seat installation**
- **Determination of occupant strike envelope**
- **Seat system modification**
- **Seat installation modification**
 - Does not include **seat-floor** attachment

Validation Acceptance

- **Validation against Dynamic Tests**
- **Use consistent with validation conditions**
- **Correlation demonstrated with**
 - Occupant trajectory
 - Floor loads
 - Restraint system loads
 - Head Injury Criterion (HIC)
 - Spine load
 - Femur compressive load

Documentation Requirements

- **Validation and Analysis Report**
- **Background and description of seat**
- **Description of model**
- **Assumptions**
- **Conditions and analysis parameters**
- **Analytical results**

SAE ARP 5765

Provide a quantitative method to measure and evaluate the degree of correlation between a model and a physical test, and to provide best modeling practices to improve the accuracy and predictability of seat analyses

SAE ARP 5765 Outline

- **v-ATD Calibration**
 - Establish v-ATD performance criteria
- **Seat System Verification and Validation**
 - How to evaluate the accuracy of seat models
- **Seat Modeling Best Practice Guide**
 - Testing
 - Modeling
- **Appendices**
 - Test-Simulation Comparison Methodology
 - Dataset for Hybrid II
 - Dataset for FAA-Hybrid III

v-ATD Calibration

- **Industry lacked multiple v-ATDs that produced accurate results in aviation-specific scenarios.**
- **Goal: define the process for ensuring that v-ATDs match the anthropometry and kinematic performance of a physical ATD for aviation-specific applications**
 - Component Response (head, chest, knee, etc.)
 - Pelvic Shape Evaluation (cushion interaction)
 - Dynamic Response
 - Forward Facing (FF) 2pt, FF 3pt, FF 4pt, Download
 - Appendices B & C: NIAR Test Information

System Validation

- **Ensure that the system (v-ATD, restraints, seat, etc.) behaves in a predictable manner**
- **Verification**
 - Code
 - Calculation
- **Validation**
 - Materials
 - Component Test
 - Sensitivity Analysis
- **Documentation**

Best Practices: Testing

- **Provide guidance for things to consider during test setup so that the necessary information is collected to support modeling**
- **Modifications to typical test protocols to collect quality data for modeling**
 - Consistent ATD positioning
 - Joint locations, pelvic angle
 - Belt information (material props, lengths, pre-tension)
 - Tips for photometric analysis
 - H-pt
 - Knee

Best Practices: Modeling

- **Information needed to build aircraft specific models**
 - Global Parameters
 - Physical Discretization
 - Material Definition (metals, foam, plastic)
 - Contact Definition
 - Load Application
 - Initial Conditions (ATD positioning, pitch and roll)
 - Output Control

Summary

- **Occupant safety through seat testing**
- **M&S can be used for seat certification**
- **Can support full airframe crashworthiness**
- **Testing will still be required**

